# RADIATIVE ENERGY TRANSFER

# 1 The Electromagnetic Spectrum

•The relation of between wavelength and frequency of electromagnetic radiation

$$\begin{pmatrix} \lambda \\ wavelength \end{pmatrix} \begin{pmatrix} v \\ Frequency \end{pmatrix} = \begin{pmatrix} C \\ Speed of Light \end{pmatrix}$$
(1)



Figure 1. The electromagnetic spectrum.

# The Electromagnetic Spectrum



relation of reflectivity, Absorptivity and Transmissivity

$$\begin{pmatrix} \rho \\ \text{Reflectivity} \end{pmatrix} + \begin{pmatrix} \alpha \\ \text{Absorptivity} \end{pmatrix} + \begin{pmatrix} \tau \\ \text{Transmissivity} \end{pmatrix} = 1 \quad (2)$$





#### Transmissivity of a Leaf Photosynthesis



Figure 4. Representative spectral values for the reflectivity, absorptivity, and transmissivity of a leaf.

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•A shaded leaf  $\rightarrow$  get radiation between 0.7~2  $\mu m$ 



Transmissivity of the Atmosphere: Greenhouse Effect



- •Greenhouse Effect -The atmosphere traps heat.
- •Greenhouse gases → Infrared: opaque
- →Atmosphere: Transparent
  - The radiation from earth by the atmosphere.

◆Greenhouse gases
<sup>-</sup>CO<sub>2</sub>, methane, chlorofluo rocarbons

Increasing its temperature.

Albedo-Reflection from Soil



Figure 5. Albedo of various surfaces.

Absorption and Transmission in Biomaterials

Beer-Lambert law : 
$$F = F_0 e^{-x/\delta}$$
 (3)

F :The flux in entering a material at its surface.

 $F_0$  :The flux at a location inside the material at a distance x from surface.

 $\delta$  :the penetration depth.



# 3 Thermal Radiation from an Ideal (Black) Body

-E :Total emissive power of a surface  $-E_{h}$  :The emissive of an ideal (Black) body  $-E_{\lambda}^{\nu}$ : The emissive power of a wavelength  $\lambda$ The monochromatic emissive power  $dE = E_{\lambda}d\lambda$  $\longrightarrow$  Energy emitted between  $\lambda_1$  and  $\lambda_2$  $E = \int_{\lambda}^{\lambda_2} E_{\lambda} d\lambda$ Total energy emitted by a body can be written as:  $E = \int_{0}^{\infty} E_{\lambda} d\lambda$ Fraction of total energy emitted two wavelengths  $F = \frac{\int_{\lambda_1}^{\lambda_2} E_{\lambda} d\lambda}{\int_{0}^{\infty} E_{\lambda} d\lambda}$ (5)

Thermal Radiation from an Ideal (Black) Body

$$E_{b,\lambda} = \frac{2\pi c^2 h \lambda^{-5}}{\exp(ch/\kappa\lambda T) - 1}$$
(6)

- C : The speed of light.
- H : The plank's constant.
- $\kappa$ : The Boltzmann's constant.

Total energy emitted at all wavelengths

$$E_{b} = \int E_{b,\lambda} d\lambda = \frac{2\pi^{5} \kappa^{4} T^{4}}{15c^{2}h^{3}} = \sigma T^{4} \qquad (7)$$

#### 4 Fraction of Energy Emitted by an Ideal Body in a Given Range of Wavelengths



Energy emitted between  $\lambda_1$  and  $\lambda_2$ 

Figure 7. Illustration of energy emitted by a blackbody between two wavelengths.

Fraction of the total energy emitted by the same blackbody

(8)

$$\frac{\int_{\lambda_{1}}^{\lambda_{2}} E_{b,\lambda} d\lambda}{\int_{0}^{\infty} E_{b,\lambda} d\lambda}$$

#### Fraction of Energy Emitted by an Ideal Body in a Given Range of Wavelengths

$$= \frac{\int_{0}^{\lambda_{2}} \mathcal{E}_{b,\lambda} d\lambda - \int_{0}^{\lambda_{1}} \mathcal{E}_{b,\lambda} d\lambda}{\sigma T^{4}}$$
$$= \int_{0}^{\lambda_{2}} \frac{\mathcal{E}_{b,\lambda}}{\sigma T^{4}} d\lambda - \int_{0}^{\lambda_{1}} \frac{\mathcal{E}_{b,\lambda}}{\sigma T^{4}} d\lambda$$
$$= \int_{0}^{\lambda_{2}} \frac{\mathcal{E}_{b,\lambda}}{\sigma T^{5}} d(\lambda T) - \int_{0}^{\lambda_{1}} \frac{\mathcal{E}_{b,\lambda}}{\sigma T^{5}} d(\lambda T) \quad (9)$$

-substituting Eqn.8.6 for the integrand in the two integrals

$$\frac{E_{b,\lambda}}{\sigma T^5} = \frac{2\pi c^2 h(\lambda T)^{-5}}{\exp(ch/\kappa\lambda T) - 1}$$
(10)

#### Fraction of Energy Emitted by an Ideal Body in a Given Range of Wavelengths



$$F_{0-\lambda T} = \int_{0}^{\lambda T} \frac{E_{b,\lambda}}{\sigma T^{5}} d(\lambda T) \quad (11)$$



Figure 8. Graphical illustration of  $F_{0-\lambda T}$ 



Figure 9. Graphical illustration of  $F_{0-\lambda,T} - F_{0-\lambda,T}$  .

# 5 Thermal radiation from a real body : Emissivity

Total emissivity

$$\epsilon = \frac{E}{E_b}$$
 where  $0 \le \epsilon \le 1$  (12)



Figure 10. Emissivity of biomaterials as compared to other materials such as aluminum and water.

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[Emissivity]
Polished metals < nonmetal surfaces < biomaterials
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Thermal radiation from a real body : Emissivity

The monochromatic emissivity

: 
$$E_{\lambda} = \frac{E_{\lambda}}{E_{b,\lambda}}$$
 where  $0 \le \epsilon \le 1$  (13)

#### **&** Gray body :

The body which is independent of wave length for monochromatic emissivity



An average value of 0.98–0.99 for human skin monochromatic emissivity is often used.

Figure 11.The spectral emissivity of human skin at room temperature. The corresponding total emittance is 0.993. Most authors use values between 0.98 and 0.99.

6 Emission from human bodies – Infrared and microwave thermography for clinical uses

All bodies above 0 K emit radiation over a range of wavelengths. So, a human body would also emit radiation.



Figure 12. Radiation emitted from a blackbody at 300 K showing the approximate radiation emitted from humans.

- 1. Human radiation includes wavelengths from infrared to microwave region.
- 2. At a 300 K, which is near the human skin temperature, maximum intensity of radiative emission occurs at wavelengths near 10  $\mu m$ .
- 3. It is used in the medical field.

# 7 Solar, Atmospheric, and Earth surface radiation Solar radiation

Outside the atmosphere, the solar radiation reaches an approximate intensity  $E_{sc}$  of 1353  $W/m^2$ .



Figure 13. Relation of solar radiation on earth surface to the extraterrestrial solar radiation and the solar constant.

$$E_{et} = E_{sc} \cdot \beta \cdot cos\theta$$

- $E_{et} \Rightarrow$  The extraterrestrial solar irradiation  $(W/m^2)$
- $E_{sc} \Rightarrow$  the solar constant of 1353  $W/m^2$
- $\beta \Rightarrow$  A correction factor to account for the eccentricity of the earth's orbit about the sun (0.97  $\leq \beta \leq 1.03$ )
- $\theta \Rightarrow$  The zenith angle

#### Solar radiation



Figure 14. Spectral distribution of solar radiation outside the earth's atmosphere and on the earth surface.

Comparison of black body to extraterrestrial solar radiation and solar irradiation at sea level

Emission from Earth's surface

The energy emitted by the earth is given by :

$$E_{earth} = \epsilon_{earth} \sigma T_{earth}^{4} \tag{14}$$

The average temperature of the earth's surface is considered to be 14°C. The emissivity of the earth surface will be made up of that of water, soil, etc.

Atmospheric emissions

The atmospheric gases emit considerable radiation. Effective concentration of these atmospheric gases can vary from time to time.

$$G_{atm} = \epsilon_{sky} \sigma T_a^{4} \tag{15}$$

 $T_a$  : the air temperature near the ground.

The apparent sky emissivity,  $\epsilon_{sky}$ , is an empirical constant.

$$\epsilon_{sky} = 0.55 + 1.8(\frac{p_{H_20}}{P})^{1/2}$$
 (16)

 $p_{H_2O}$ : the partial pressure of water vapor in the atmosphere. *P*: the total atmospheric pressure.

#### Global energy balance



Figure 15. The global energy balance for annual mean conditions.

At the earth's surface, energy balance is

lnput - Output = (169+327) - (390+90+16) = 0

When two bodies exchange thermal radiation, the net radiative exchange between them will depend on their size, shape, and the relative orientation of their radiating surfaces.

' $F_{1-2}$ ', which is called as configuration factor or view factor, includes size, shape and orientation factor.

 $F_{1-2}$  = fraction of radiation leaving surface 1 that is intercepted by surface 2

View factor is dimensionless. The range of this is zero to one.



Figure 16. Illustration of view factors showing view factor is zero for a surface to itself for a plane or a convex surface, but non-zero for a concave surface.

- 1. The view factor from a surface to itself will be zero unless the surface "sees" itself.
- 2.  $F_{1-1} = 0$  for plane or convex surfaces and  $F_{1-1} \neq 0$  for concave surfaces.
- 3. There are two limiting case of view factor. One case of  $F_{1-2} = 0$  represents the situation when two surfaces do not have a direct view of each other. The other case  $F_{1-2} = 1$  represents the situation where surface 2 completely surrounds surface 1.

This page is method of calculating view factor. Main factors are distance(D) between two bodies and the sides of the rectangles (X and Y).

$$(\bar{X} = \frac{X}{D} \text{ and } \bar{Y} = \frac{Y}{D})$$



Figure 17. View factor between two aligned parallel rectangles of equal size.

The simplest situation of radiative exchange occurs when two blackbodies are exchanging radiation.

The net radiative exchange between two such bodies is given by

$$q_{1-2} = \sigma A_1 F_{1-2} (T_1^4 - T_2^4) \tag{17}$$

or

$$q_{2-1} = \sigma A_2 F_{2-1} (T_2^4 - T_1^4) \tag{18}$$

 $q_{1-2}(W)$ : The net energy transfer between bodies 1 and 2.  $A_1, A_2(m^2)$ : The areas of bodies.  $F_{1-2}$ : The view factor between bodies 1 and 2.  $T_1, T_2(K)$ : The absolute temperature of the respective surfaces.

Instead of blackbodies, when two gray bodies are exchanging radiation, the situation becomes more complex since these bodies are also reflecting part of the radiation.

For the special case when two gray surfaces form a complete enclosure or close to it, as illustrated in Figure 18.

The net radiative exchange is given by

$$q_{1-2} = \frac{\sigma(T_1^4 - T_2^4)}{\frac{1-\epsilon_1}{\epsilon_1 A_1} + \frac{1}{F_{1-2}A_1} + \frac{1-\epsilon_2}{\epsilon_2 A_2}}$$
(19)

 $q_{1-2}(W)$ : The net energy transfer between bodies 1 and 2.  $\epsilon_i$ : The emissivity of body *i*.  $A_i(m^2)$ : The area of body *i*.  $T_i(K)$ : The absolute temperature of surface *i*.  $F_{1-2}$ : The view factor between bodies 1 and 2.



Figure 18. Schematic of two gray surfaces that form a complete enclosure.

For example, radiative exchange between two large parallel planes, as shown in Figure 19, would be given by a special case of Eqn. 8.24 for  $A_1 = A_2 = A$  and  $F_{1-2} = 1$ .



(20)



Figure 19. Schematic of two gray infinite parallel planes.

When a relatively small object 1 is completely enclosed by a much larger surface 2,  $A_1/A_2 \approx 0$  and  $F_{1-2} = 1$ , and Eqn. 8.24 simplifies to

$$q_{1-2} = \epsilon_1 \sigma A_1 (T_1^4 - T_2^4) \tag{21}$$

#### Radiative Heat Transfer Coefficient

For a special case of radiative heat exchange, when temperatures  $T_1$  and  $T_2$  are close, Eqn 8.21 can be written as

$$q_{1-2} = \sigma A_1 F_{1-2} 4T^3 (T_1 - T_2)$$
(22)

$$= A_1 F_{1-2} h_r \left( T_1 - T_2 \right) \tag{23}$$

 $h_r(\frac{W}{m^2 \cdot K})$  is a radiative heat transfer coefficient similar to the convective heat transfer coefficient.

$$h_r = 4\sigma T_1^3 \tag{24}$$

 $h_r$  may be determined experimentally.

The radiative heat transfer coefficient was measured experimentally as:

$$h_r = 3.5[1 + 0.0055(T_s + T_a)]$$
(25)

 $T_s(K)$ : The average skin temperature.  $T_a(K)$ : The environmental temperature.

#### Radiative Exchange Between a Leaf and Surrondings



Figure 20. Schematic of radiation exchange of a leaf with the sky and the ground.

Radiative heat loss from an unshaded leaf has important consequences.

Radiative freeze typically occurs when the wind is calm or still.

So the uppermost leaves in a tress will tend to have the lowest temperatures and frost tends to form there first.

Consequently, when such freezing is imminent, large fans are employed to blow air.

#### Radiative Exchange Between Animal and Its Surroundings

A MRT(mean radiant temperature) : An average temperature of the walls.



Figure 21. Human Thermal Comfort and Mean Radiant Temperature.

1. Body heat loss by radiation is an important mechanism for maintaining a thermal balance and comfort.

2. The shaded zone for human thermal comfort as related to surrounding wall temperature shown in Fig. 21.

3. The higher the MRT value, the lower the corresponding air temperature required for comfort.



By this theory, We should maintain thermal comfort by controlling the convective heat loss.