Chapter 9

Turbulent flow and boundary layer



Random fluctuating functions of time

Characterized by mean, intensity, frequency spectrum

$$\overline{u}_z \equiv \frac{1}{\tau} \int_t^{t+\tau} u_z \, dt$$

Time averaging

$$\overline{S} \equiv \frac{1}{\tau} \int_{t}^{t+\tau} S \, dt \qquad \overline{S} \equiv \frac{1}{\tau} \int_{t}^{t+\tau} (\overline{S} + S') \, dt = \overline{\overline{S}} + \overline{S'} \qquad \overline{S'} \equiv 0$$

$$\overline{TS} \equiv \frac{1}{\tau} \int_{t}^{t+\tau} (\overline{T} + T') (\overline{S} + S') dt = \overline{TS} + \overline{T'S} + \overline{\overline{TS'}} + \overline{T'S'}$$
$$= \overline{TS} + \overline{T'S'}$$



Reynolds stress models

Newtonian

turbulent

 $T_{12} = -\mu\Delta_{12}$

$$\overline{T}_{zx} = -\mu_{t} \frac{\partial \overline{\mu}_{z}}{\partial x}$$

Turbulent viscosity parameter; not a fluid property

Prandtl's mixing length theory

$$\overline{T}_{zx} = -\rho \mathbf{l}^2 \left| \frac{\partial \overline{u}_z}{\partial x} \right| \left(\frac{\partial \overline{u}_z}{\partial x} \right)$$

Turbulent flow in a pipe

Assume; 1.mean flow is fully developed and steady state 2.mixing length is a linear function of distance from the wall

$$\overline{u}_z = \overline{u}_z(r)$$
$$\overline{u}_r = \overline{u}_\theta = 0$$

 $\overline{T}_{rz} = \rho \kappa^2 s^2 \left(\frac{\partial \overline{u}_z}{\partial s}\right)^2$

$$l = \kappa s$$
 where $s \equiv R - r$

Axial component of time-averaged Navier-Stokes equation

$$0 = \frac{\Delta \overline{P}}{L} + \mu \frac{1}{r} \frac{d}{dr} \left(r \frac{d\overline{u}_z}{dr} \right) - \frac{1}{r} \frac{d}{dr} (r\overline{T}_{rz})$$

1. Far from the wall; turbulence dominates viscous shear stress

$$\overline{T}_{rz} = \frac{\Delta \overline{P}R}{2L} \frac{r}{R} = T_{R} \frac{r}{R}$$

Time averaged shear stress at wall

$$\overline{T}_{rz} = \rho \kappa^2 s^2 \left(\frac{\partial \overline{u}_z}{\partial s}\right)^2 \qquad \rho \kappa^2 s^2 \left(\frac{\partial \overline{u}_z}{\partial s}\right)^2 = T_R \left(1 - \frac{s}{R}\right)$$

$$\frac{\partial \overline{u}_z}{\partial s} = \left(\frac{T_R}{\rho \kappa^2}\right)^{1/2} \frac{(1 - \underline{s}/R)^{1/2}}{s}$$

$$\overline{\int \frac{\partial \overline{u}_z}{\partial s}} = \left(\frac{T_R}{\rho \kappa^2}\right)^{1/2} \frac{(1 - \underline{s}/R)^{1/2}}{s}$$

$$\overline{\int \frac{\partial \overline{u}_z}{\partial s}} = \frac{s_1 - \frac{\text{Turbulent core}}{\text{Buffer zone}}}{\frac{1}{\text{Laminar sublayer}}} s^{-axis}$$

Solid wal

Integrating from $s=s_1$ to $s>s_1$

$$u^{+} = u_{1}^{+} + \frac{1}{\kappa} \ln \frac{s^{+}}{s_{1}^{+}} \qquad u^{+} \equiv \frac{\overline{u}_{z}}{u_{*}} \quad \text{and} \quad s^{+} \equiv \frac{su_{*}\rho}{\mu} \qquad u_{*} \equiv \left(\frac{T_{R}}{\rho}\right)^{1/2}$$
$$u^{+} = u_{1}^{+} - \frac{1}{\kappa} \ln s_{1}^{+} + \frac{1}{\kappa} \ln s^{+}$$
$$= A + B \ln s^{+} \qquad u^{+} = 2.78 \ln s^{+} + 3.6$$

2. Near the wall; viscous effect dominates

$$0 = \frac{\Delta \overline{P}}{L} + \mu \frac{1}{r} \frac{d}{dr} \left(r \frac{d\overline{u}_z}{dr} \right) - \frac{1}{r} \frac{d}{dr} (r\overline{T}_{rz})$$

$$\overline{T}_R = \frac{\Delta \overline{P}R}{2L} = \mu \frac{d\overline{u}_z}{ds} \Big|_{s=0}$$

$$1 = \frac{du^+}{ds^+} \Big|_{s=0} \qquad u^+ = s^+$$

$$0 = \frac{du^+}{10} = \frac{u^+}{10} = s^+$$

$$0 = \frac{10^+}{10^+} = \frac{10^+}{10^+} = s^+$$

Turbulent friction factor

$$f = \frac{T_{R}}{\frac{1}{2}\rho U^{2}} = \frac{\Delta \overline{P}R}{\rho U^{2}L} \qquad u_{*}^{2} = \frac{T_{R}}{\rho} = \frac{fU^{2}}{2} \qquad \frac{u_{*}^{2}}{U^{2}} = \frac{f}{2}$$

$$\pi R^{2}U = \int_{0}^{R} 2\pi r \overline{u}_{z} dr$$

$$u^{+} = 2.78 \ln s^{+} + 3.6$$

$$\frac{1}{\sqrt{f}} = 1.97 \ln(\text{Re}\sqrt{f}) - 2.45 = 4.52 \log(\text{Re}\sqrt{f}) - 2.45$$
Neglected buffer zone, laminar sublayer
$$\frac{1}{\sqrt{f}} = 4 \log(\text{Re}\sqrt{f}) - 0.4$$
Semi-empirical, but
works well

Friction factor for pipe flow

Laminar
$$\frac{\Delta P}{L} = \frac{8\mu Q}{\pi R^4} = \frac{8\mu U}{R^2} \qquad f_{\text{lam}} = \frac{\Delta P R}{\rho U^2 L} = \frac{8\mu}{\rho U R} = \frac{16}{\text{Re}}$$
Hagen-Poiseuille
Turbulent
$$\frac{1}{\sqrt{f}} = 4\log(\text{Re}\sqrt{f}) - 0.4 \qquad f_{\text{lam}} = 0.079 \,\text{Re}^{-0.25}$$



Laminar boundary layer

No matter how turbulent the flow is far from the surface



Negligible viscous effect outside the boundary layer



$$\frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0 \qquad \rho \left(u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} \right) = \mu \frac{\partial^2 u_x}{\partial y^2}$$

Boundary conditions

$$u_{x} = u_{y} = 0 \quad \text{on} \quad y = 0, \text{ for } x > 0$$
$$u_{x} = U \quad \text{for} \quad x < 0$$
$$u_{x} \rightarrow U \quad \text{for} \quad y \rightarrow \infty \qquad \frac{\partial}{\partial t}$$

$$\frac{\partial \psi}{\partial y} = -\frac{\partial \psi}{\partial x} = 0 \quad \text{on} \quad y = 0, \text{ for } x > 0$$
$$\frac{\partial \psi}{\partial y} \to U \quad \text{for} \quad x < 0, \text{ and } y \to \infty$$

Blasius' similarity transformation

$$\Psi = \frac{\psi}{(U\nu x)^{1/2}} \qquad \eta \equiv y \left(\frac{U}{\nu x}\right)^{1/2} \qquad \boxed{2\Psi''' + \Psi\Psi'' = 0} \qquad \qquad \Psi = \Psi' = 0 \quad \text{at} \quad \eta = 0$$
$$\Psi' \to 1 \quad \text{as} \quad \eta \to \infty$$

$$\frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2} = v \frac{\partial^3 \psi}{\partial y^3}$$





 U_x becomes nearly Ualong the boundary layer $y = \delta(x)$

$$\delta \left(\frac{U}{vx}\right)^{1/2} = 5$$

Boundary layer thickness

Drag coeff.

$$C_{\rm D} \equiv \frac{-\mu(\partial u_x/\partial y)_{y=0}}{\frac{1}{2}\rho U^2} \equiv 0.664 \left(\frac{\nu}{xU}\right)^{1/2} = \frac{0.664}{(\text{Re}_x)^{1/2}}$$
Shear force

$$F_{\rm s} = W \int_0^L -\mu \left(\frac{\partial u_x}{\partial y}\right)_{y=0} dx \qquad \frac{F_{\rm s}/LW}{\frac{1}{2}\rho U^2} \equiv \overline{C}_{\rm D} = \frac{1.328}{(UL/\nu)^{1/2}} = \frac{1.328}{(\text{Re}_{\rm L})^{1/2}}$$
Local Reynolds
number

$$\text{Re}_x \equiv \frac{xU}{\nu}$$

$$\text{Re}_x^{\text{trans}} = 5 \times 10^5 \quad \text{(transition to turbulence)}$$

Turbulent drag force

$$f = C_{\rm D} = \frac{\text{total force}}{\frac{1}{2}\rho U^2 \pi R^2}$$





Associated with transitions in the character of the boundary layer along the spherical surface