CHBE320 LECTURE IX FREQUENCY RESPONSES

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Fall 2021

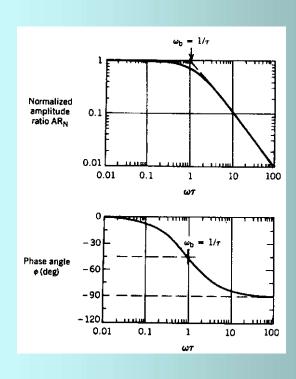
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Road Map of the Lecture IX

Frequency Response

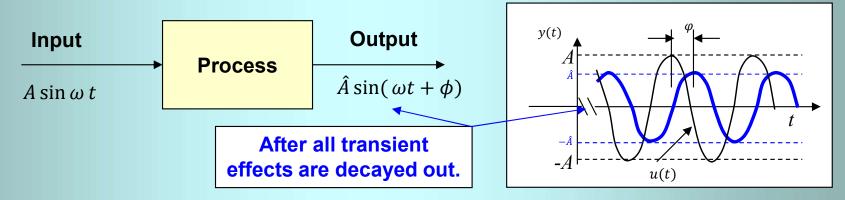
- Definition
- Benefits of frequency analysis
- How to get frequency response
- Bode Plot
- Nyquist Diagram



DEFINITION OF FREQUENCY RESPONSE

For linear system

- "The ultimate output response of a process for a sinusoidal input at a frequency will show amplitude change and phase shift at the same frequency depending on the process characteristics."



- Amplitude ratio (AR): attenuation of amplitude, \hat{A}/A
- Phase angle (ϕ): phase shift compared to input
- These two quantities are the function of frequency.

BENEFITS OF FREQUENCY RESPONSE

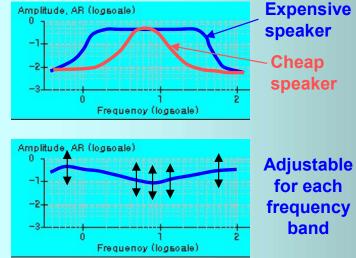
- Frequency responses are the informative representations of dynamic systems
 - Audio Speaker

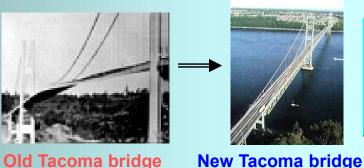


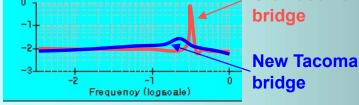
Equalizer



Structure





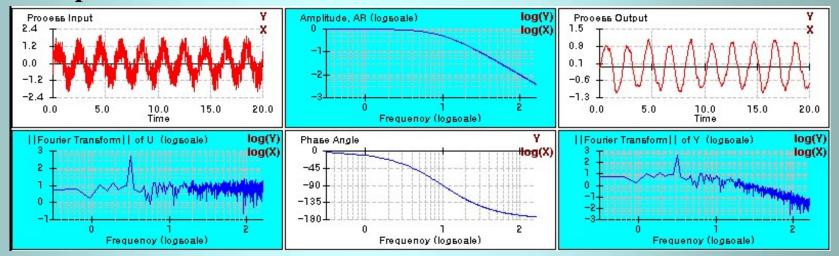


Programmy (logacia)

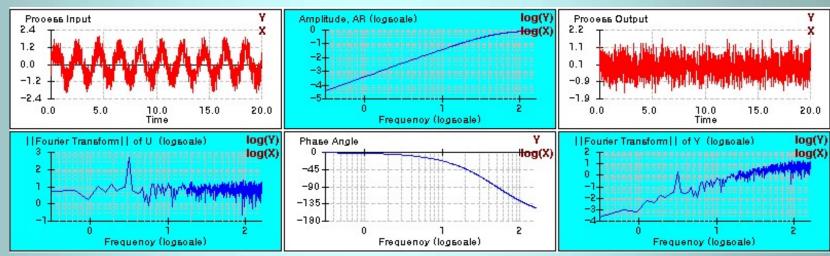
Amplitude, AR (logsoale)

Old Tacoma

Low-pass filter



High-pass filter

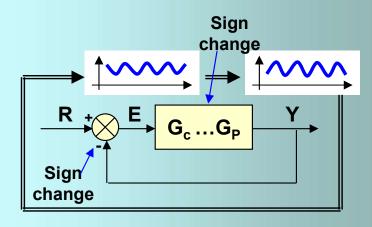


In signal processing field, transfer functions are called "filters".

- Any linear dynamical system is completely defined by its frequency response.
 - The AR and phase angle define the system completely.
 - Bode diagram
 - AR in log-log plot
 - Phase angle in log-linear plot
 - Via efficient numerical technique (fast Fourier transform,
 FFT), the output can be calculated for any type of input.
- Frequency response representation of a system dynamics is very convenient for designing a feedback controller and analyzing a closed-loop system.
 - Bode stability
 - Gain margin (GM) and phase margin (PM)

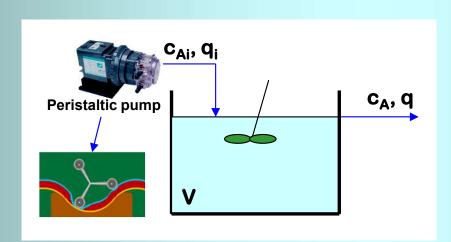
Critical frequency

- As frequency changes, the amplitude ratio (AR) and the phase angle (PA) change.
- The frequency where the PA reaches –180° is called critical frequency (ω_c).
- The component of output at the critical frequency will have the exactly same phase as the signal goes through the loop due to comparator (-180°) and phase shift of the process (-180°).
- For the open-loop gain at the critical frequency, $K_{OL}(\omega_c) = 1$
 - No change in magnitude
 - Continuous cycling
- **For** $K_{OL}(\omega_c) > 1$
 - Getting bigger in magnitude
 - Unstable
- **For** $K_{OL}(\omega_c) < 1$
 - Getting smaller in magnitude
 - Stable



Example

If a feed is pumped by a peristaltic pump to a CSTR, will the fluctuation of the feed flow appear in the output?



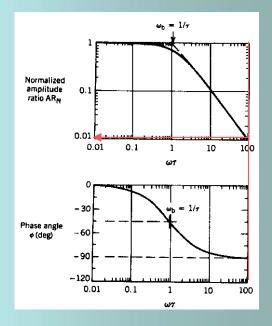


$$V\frac{dc_A}{dt} = q_i c_{Ai} - q c_A \ (q \approx \text{constant})$$

$$\frac{C_A(s)}{q_i(s)} = \frac{C_{Ai}}{Vs + q} = \frac{C_{Ai}/q}{(V/q)s + 1}$$

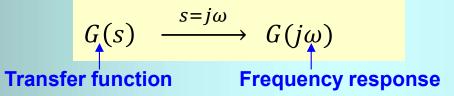
- V=50cm³, q=90cm³/min (so is the average of q_i)
 - Process time constant=0.555min.
- The rpm of the peristaltic pump is 60rpm.
 - Input frequency=180rad/min (3blades)
- The AR=0.01 (ωτ = 100)

If the magnitude of fluctuation of q_i is 5% of nominal flow rate, the fluctuation in the output concentration will be about 0.05% which is almost unnoticeable.



OBTAINING FREQUENCY RESPONSE

• From the transfer function, replace s with $j\omega$



- For a pole, $s = \alpha + j\omega$, the response mode is $e^{(\alpha+j\omega)t}$.
- If the modes are not unstable ($\alpha \le 0$) and enough time elapses, the survived modes becomes $e^{j\omega t}$. (ultimate response)
- The frequency response, $G(j\omega)$ is complex as a function of frequency.

$$G(j\omega) = \text{Re}[G(j\omega)] + j \text{Im}[G(j\omega)]$$

$$AR = |G(j\omega)| = \sqrt{\text{Re}[G(j\omega)]^2 + \text{Im}[G(j\omega)]^2}$$

$$\phi = \angle G(j\omega) = \tan^{-1}(\text{Im}[G(j\omega)]/\text{Re}[G(j\omega)])$$

▶Bode plot

Nyquist

Getting ultimate response

- For a sinusoidal forcing function $Y(s) = G(s) \frac{A\omega}{s^2 + \omega^2}$

$$Y(s) = G(s) \frac{A\omega}{s^2 + \omega^2}$$

- Assume G(s) has stable poles b_i .

Decayed out at large t

$$Y(s) = G(s)\frac{A\omega}{s^2 + \omega^2} = \frac{\alpha_1}{s + b_1} + \dots + \frac{\alpha_n}{s + b_n} + \frac{Cs + D\omega}{s^2 + \omega^2}$$
$$G(j\omega)A\omega = Cj\omega + D\omega \Rightarrow G(j\omega) = \frac{D}{A} + j\frac{C}{A} = R + jI$$

$$C = IA, D = RA \Rightarrow y_{ul} = A(I\cos\omega t + R\sin\omega t) = \hat{A}\sin(\omega t + \phi)$$

$$\therefore AR = \hat{A}/A = \sqrt{R^2 + I^2} = |G(j\omega)| \quad \text{and} \quad \phi = \tan^{-1}(I/R) = \measuredangle G(j\omega)$$

- Without calculating transient response, the frequency response can be obtained directly from $G(j\omega)$.
- Unstable transfer function does not have a frequency response because a sinusoidal input produces an unstable output response.

First-order process

$$G(s) = \frac{K}{(\tau s + 1)}$$

$$G(j\omega) = \frac{K}{(1+j\omega\tau)} = \frac{K}{(1+\omega^2\tau^2)}(1-j\omega\tau)$$

$$AR_N = |G(j\omega)| = \frac{1}{\sqrt{1 + \omega^2 \tau^2}}$$

$$\phi = \measuredangle G(j\omega) = -\tan^{-1}(\omega\tau)$$

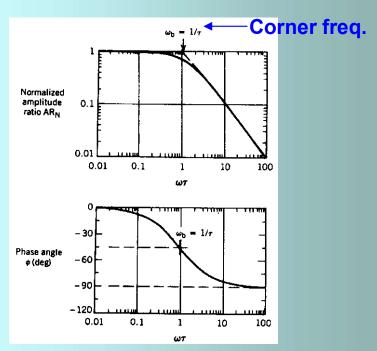
Second-order process

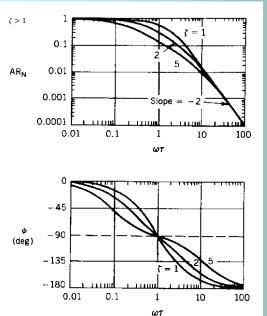
$$G(s) = \frac{K}{(\tau^2 s^2 + 2\zeta \tau s + 1)}$$

$$G(j\omega) = \frac{K}{(1 - \tau^2 \omega^2) + 2j\zeta\tau\omega}$$

$$AR = |G(j\omega)| = \frac{K}{\sqrt{(1 - \omega^2 \tau^2)^2 + (2\zeta\omega\tau)^2}}$$

$$\phi = \angle G(j\omega) = \tan^{-1} \frac{\operatorname{Im}(G(j\omega))}{\operatorname{Re}(G(j\omega))} = -\tan^{-1} \frac{2\zeta\omega\tau}{1 - \omega^2\tau^2}$$





Process Zero (lead)

$$G(s) = \tau_a s + 1$$

$$G(j\omega) = 1 + j\omega\tau_a$$

$$AR_N = |G(j\omega)| = \sqrt{1 + \omega^2 \tau_a^2}$$

$$\phi = \angle G(j\omega) = \tan^{-1}(\omega \tau_a)$$

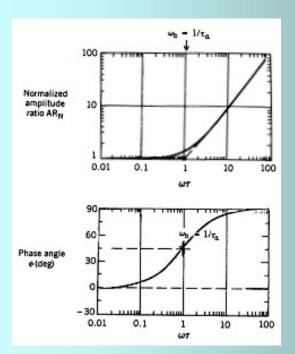
Unstable pole

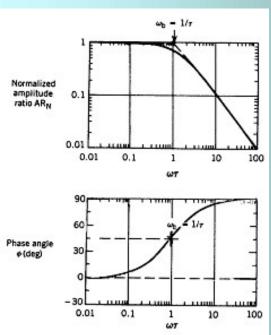
$$G(s) = \frac{1}{(-\tau s + 1)}$$

$$G(j\omega) = \frac{1}{1 - i\tau\omega} = \frac{1}{1 + \tau^2\omega^2} (1 + j\tau\omega)$$

$$AR = |G(j\omega)| = \frac{1}{\sqrt{1 + \omega^2 \tau^2}}$$

$$\phi = \measuredangle G(j\omega) = \tan^{-1} \frac{\operatorname{Im}(G(j\omega))}{\operatorname{Re}(G(j\omega))} = \tan^{-1} \omega \tau$$





Integrating process

$$G(s) = \frac{1}{As}$$
 $G(j\omega) = \frac{1}{jA\omega} = -\frac{1}{A\omega}j$

$$AR_N = |G(j\omega)| = \frac{1}{A\omega}$$

$$\phi = \measuredangle G(j\omega) = \tan^{-1}(-\frac{1}{0 \cdot \omega}) = -\frac{\pi}{2}$$

Differentiator

$$G(s) = As$$
 $G(j\omega) = jA\omega$

$$AR_N = |G(j\omega)| = A\omega$$

$$\phi = \angle G(j\omega) = \tan^{-1}(\frac{1}{0 \cdot \omega}) = \frac{\pi}{2}$$

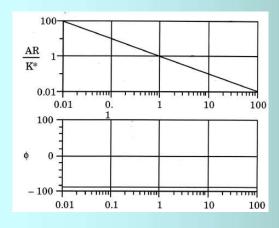
Pure delay process

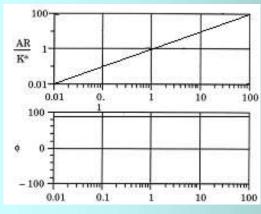
$$G(s) = e^{-\theta s}$$

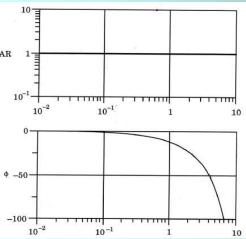
$$G(j\omega) = e^{-j\theta\omega} = \cos\theta \,\omega - j\sin\theta \,\omega$$

$$AR = |G(j\omega)| = 1$$

$$\phi = AG(j\omega) = -\tan^{-1}\tan\theta \ \omega = -\theta\omega$$







SKETCHING BODE PLOT

$$G(s) = \frac{G_a(s)G_b(s)G_c(s)\cdots}{G_1(s)G_2(s)G_3(s)\cdots}$$

$$G(j\omega) = \frac{G_a(j\omega)G_b(j\omega)G_c(j\omega)\cdots}{G_1(j\omega)G_2(j\omega)G_3(j\omega)\cdots}$$

$$|G(j\omega)| = \frac{|G_a(j\omega)||G_b(j\omega)||G_c(j\omega)|\cdots}{|G_1(j\omega)||G_2(j\omega)||G_3(j\omega)|\cdots}$$

Bode diagram

- AR vs. frequency in log-log plot
- PA vs. frequency in semi-log plot
- Useful for
 - Analysis of the response characteristics
 - Stability of the closed-loop system only for open-loop stable systems with phase angle curves exhibit a single critical frequency.

Amplitude Ratio on log-log plot

- Start from steady-state gain at $\omega = 0$. If G_{OL} includes either integrator or differentiator it starts at ∞ or 0.
- Each first-order lag (lead) adds to the slope –1 (+1) starting at the corner frequency.
- Each integrator (differentiator) adds to the slope –1 (+1) starting at zero frequency.
- A delays does not contribute to the AR plot.

Phase angle on semi-log plot

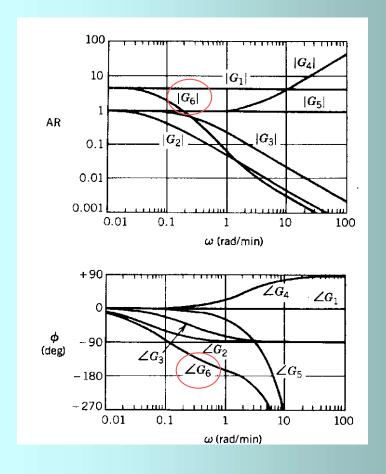
- Start from 0° or -180° at $\omega = 0$ depending on the sign of steady-state gain.
- Each first-order lag (lead) adds 0° to phase angle at $\omega = 0$, adds -90° (+90°) to phase angle at $\omega = \infty$, and adds -45° (+45°) to phase angle at corner frequency.
- Each integrator (differentiator) adds -90° (+90°) to the phase angle for all frequency.
- A delay adds $-\theta\omega$ to phase angle depending on the frequency.

Examples

1.
$$G(s) = \frac{K}{(10s+1)(5s+1)(s+1)}$$
_{G₁}
_{G₂}
_{G₃}

AR
$$0.01$$
 0.0001
 0.0001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
 0.001
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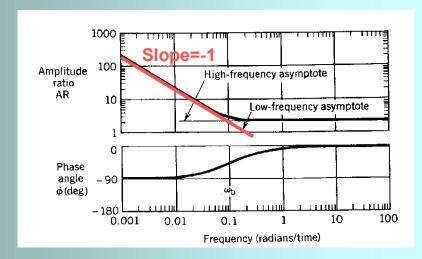
2.
$$G(s) = \frac{{{\mathbf{G_1}} \quad {\mathbf{G_4}} \quad {\mathbf{G_5}}}}{{5(0.5s+1)e^{-0.5s}}}$$



ω (rad/min)

3. PI:
$$G(s) = K_C \left(1 + \frac{1}{\tau_I s} \right)$$

5. PID:
$$G(s) = K_C \left(1 + \frac{1}{\tau_I s} + \tau_D s \right)$$

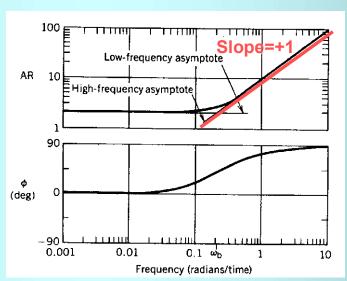


$$\omega_b = 1/\tau_I \text{ at } \phi = -45^\circ$$

$$\omega_{Notch} = \frac{1}{\sqrt{\tau_I \tau_D}}$$
 at $\phi = 0^\circ$

4. PD:
$$G(s) = K_C(1 + \tau_D s)$$

$$\omega_b = 1/\tau_D$$
 at $\phi = 45^\circ$



NYQUIST DIAGRAM

- Alternative representation of frequency response
- Polar plot of $G(j\omega)$ (ω is implicit)

$$G(j\omega) = \text{Re}[G(j\omega)] + j \text{Im}[G(j\omega)]$$

- Compact (one plot)
- Wider applicability of stability analysis than Bode plot
- High frequency characteristics will be shrunk near the origin.
 - Inverse Nyquist diagram: polar plot of $1/G(j\omega)$
- Combination of different transfer function components is not easy as with Nyquist diagram as with Bode plot.

